

Space Crank Application

Designers at Mark Systems Incorporated ran into a vexing mechanical design problem which involved moving a lens into and out of an optical plane. They suggested a number of possible solutions, none of which satisfied the project engineer. The author of this case study was called in as a consultant; details of the problem and of his solution to it are documented on the following pages.

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Prepared by Mr. Aaron Baumgarten with support from the National Science Foundation through the Case Program, Design Division, Department of Mechanical Engineering.

Mark Systems Incorporated is a young Santa Clara company whose principal business has been the manufacture of portable film processing equipment. In order to insure the company's continuing existence and growth, the management of Mark Systems has been eager to develop other products.

Early in 1966 MSI management learned of a unique engineering concept which solved in principle a difficult problem in the field of optical equipment. A good, practical conversion of this solution to functioning hardware would be of great interest to both civilian and military users. The marketing people at MSI conducted a survey intended to gauge the total market. Their estimates of the total market potential were so great that even after allowing for normal "Salesman's Enthusiasm" MSI management felt justified in making the decision to develop a working model for demonstration purposes.

The device chosen to demonstrate the basic concept was a special binocular; it was a relatively crude device, but it sufficed to show that the basic idea was a good one. Proof of the effectiveness of the demonstrator was provided by the fact that MSI began receiving orders for production units. Undoubtedly, among the users of the production unit would be sportsmen and military people holding the device in their hands while working under adverse field conditions.

A group of MSI designers was assigned the task of developing from the demonstration model a design suitable for production; they quickly ran into several nasty mechanical design problems. One of these was the problem of moving a lens into and out of an optical train by the use of the left thumb or a finger squeeze of the left hand, possibly while wearing heavy gloves (see Exhibit 1 for a visualization of this problem). The project engineer for this contract was young, aggressive, creative, and not easily pleased with marginal solutions. His designers offered slotted links, cable drives, bell cranks, spur and bevel gear trains, and cams (see Exhibits 1,2,3), but none of these satisfied him.

His objections were that there were too many pieces (expensive); or too much friction and weight (a military man, for example, would throw the device away if it were difficult to use); or too delicate or not positive (reliability is important!). He kept reminding his designers of the design criteria he had established for this particular problem. The mechanism had to:

1. have as few parts as possible
2. be light weight and compact
3. be positive in its action
4. be smooth acting and free of jamming
(easily operable to avoid operator fatigue)
5. be relatively insensitive to shock
6. be inexpensive to fabricate
7. have as little wear as possible

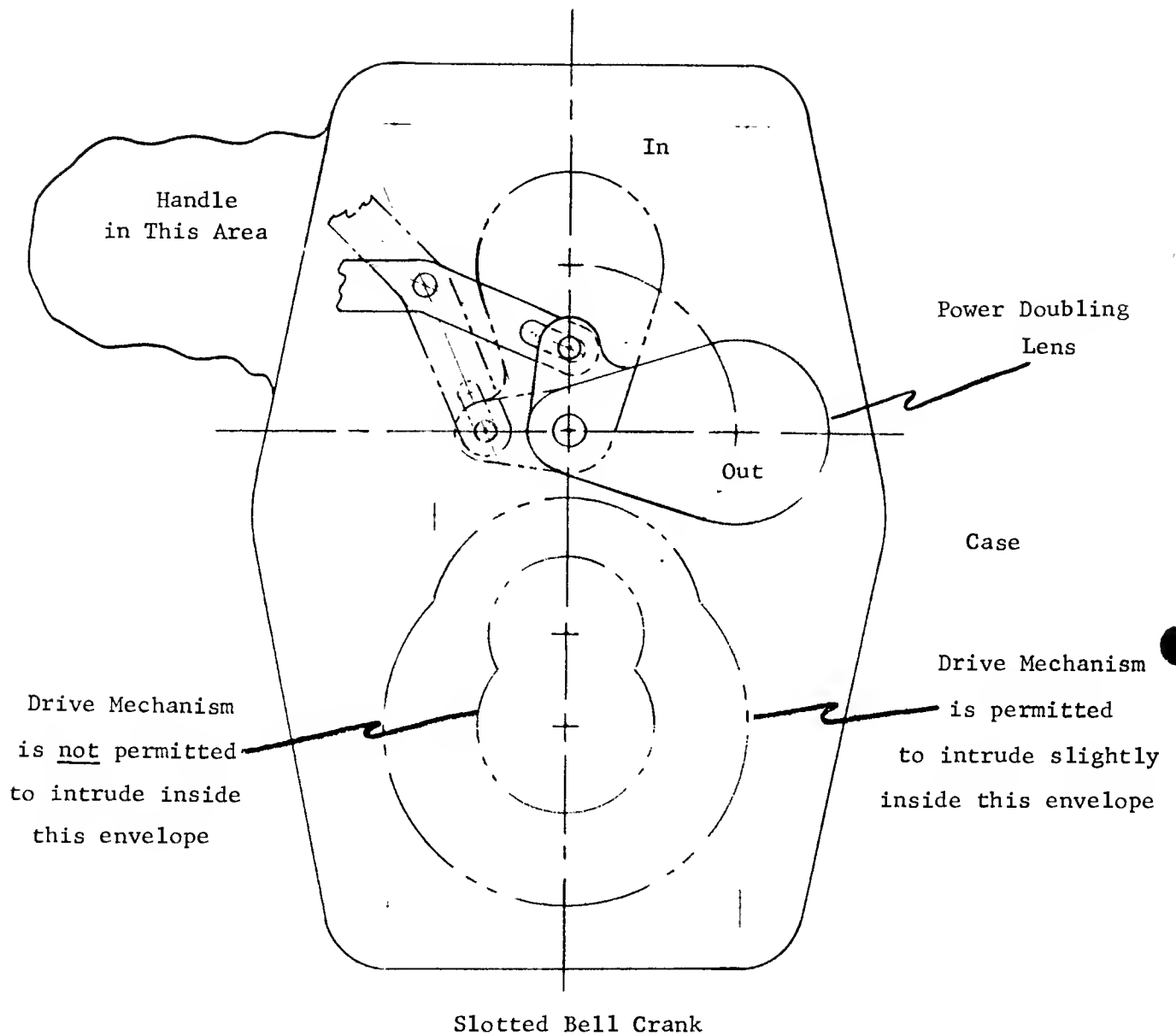
The designers replied that after all the ideas they had discussed, there didn't seem to be anything else to try. With an eye on his schedule and budget, the project engineer decided to invest in some outside consulting help from the author of this case history. The decision was based on the fact that the author had participated in the very early work on the demonstrator and was not new to its problems. Also, all of the designers knew him and would accept him readily. Twenty working hours were allowed to present a feasible concept. The author listened to the story and commented, "Unfortunately, I can agree with all of you. None of the proposed solutions looks good, but what else is there to try?" The project engineer's answer was, "Don't ask us, you're being paid to tell us what else there is to try." After an embarrassing silence, the author said, "Wait a minute. I'm not certain, but there may be something. I remember a mechanism called a space crank - some of the modern text books on kinematics describe it, and if my memory serves me correctly there was an article in a trade journal some time ago which discussed this mechanism. I've never used it before but I think it holds a lot of promise for your application. I'll look through my file of old magazine articles to see if I can find something on it." The search did turn up an article on the subject of space cranks. From the description, it seemed to fit the operating and fabricating requirements quite well. For example, given a

uniform input, the linkage output characteristics (displacement, velocity, and acceleration) were quite smooth and free of discontinuities. This indicated smooth action which would be free of jamming. The displacement curve, in particular, displayed a very interesting and desirable shape; in the region of extreme angular displacement this curve became quite flat in a manner somewhat like a "dwell". This in turn meant that it would not be necessary to develop a precise input motion in order to achieve a reasonably precise output motion. Furthermore, in this dwell condition the linkage formed a natural positive acting detent, i.e., it could not be driven by its load. Also, the manufacture of links joined by pins is relatively simple and inexpensive (especially inexpensive when considering production quantities which might number in the thousands. In terms of cost the closest competitor would be catalog gears, but even they would cost more than the links.) It seemed as though the links could be small (i.e., light) and yet rugged (i.e., dependable and insensitive to shock). For these reasons a layout was prepared which showed that the space linkage approach was indeed feasible as well as attractive from a functional and manufacturing point of view.

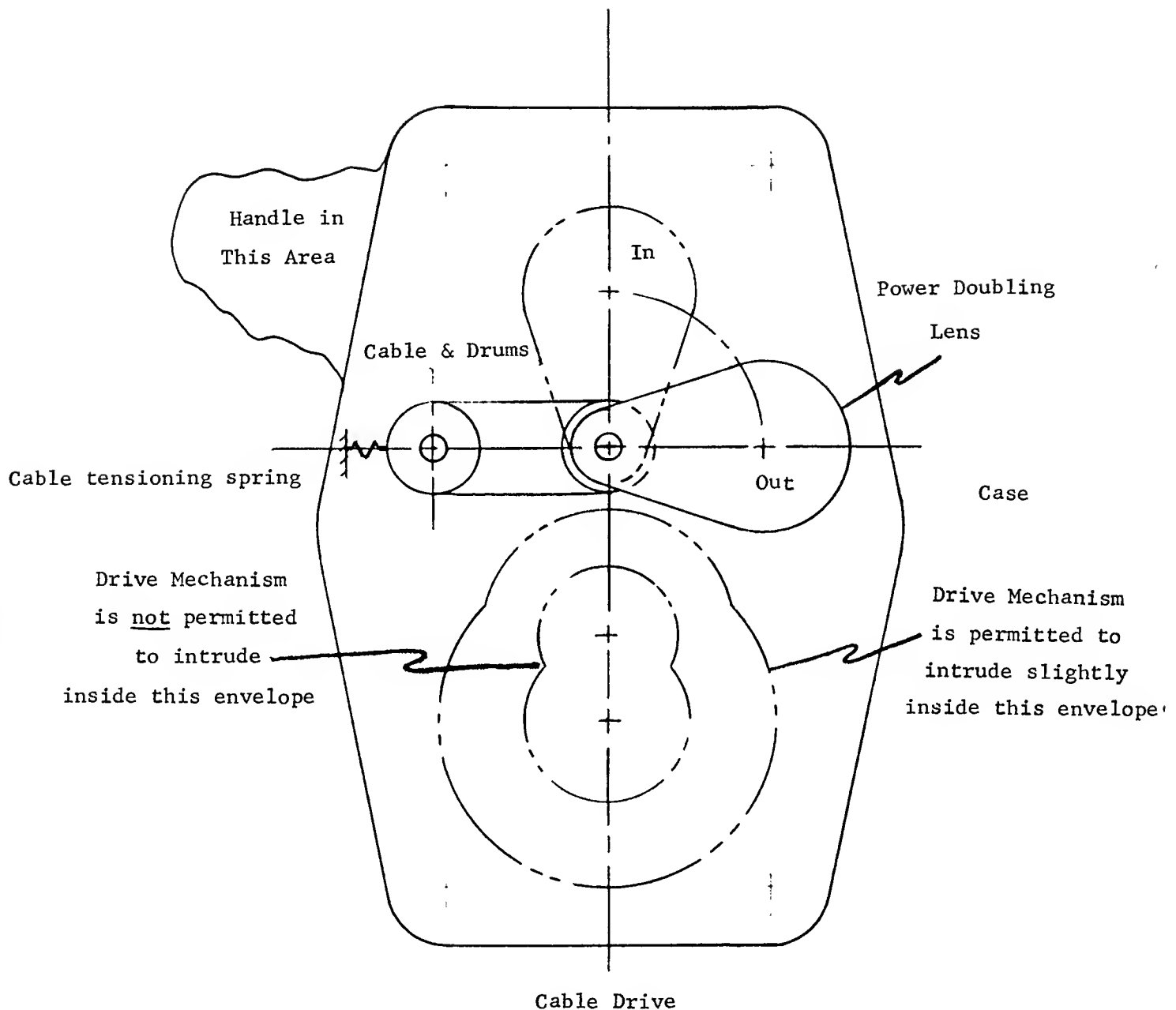
However, in order to build a prototype of the space crank so that its performance could be verified, it was necessary to establish the values of the angles which enter into the design equations (see Exhibit 4). Inasmuch as the desired maximum output angle was known ($\phi_{\max} = 90^\circ$) and the maximum output angle is a simple function of the angle called gamma ($\phi_{\max} = 2 \text{ gamma}$), part of the problem was already solved. The design equations show, however, that the angle lambda influences the displacement, velocity, and acceleration characteristics of the crank output. The details of this influence were not immediately obvious to the author from an examination of the equations. In addition, some constraint on the range of permissible values was inherent in the space available (see Exhibit 6). This was especially important because of the dwell characteristic of the displacement curves in the region of maximum output angle. The available information indicated that a desirable displacement characteristic could be achieved by using small values for the angle lambda; but this information was insufficient for a design decision on a production model.

Somehow the author had to investigate this aspect of the crank design in order to develop a "feeling" for how it would behave. There was nothing available in what remained of his original twenty hours except to attempt a manual solution of the equations for several representative values of λ and γ . The calculations were done with the aid of a desk-top calculator (how welcome a computer would have been!) A sample of the format used for this work is shown as Exhibit 5. Curves showing typical results are Exhibit 7.

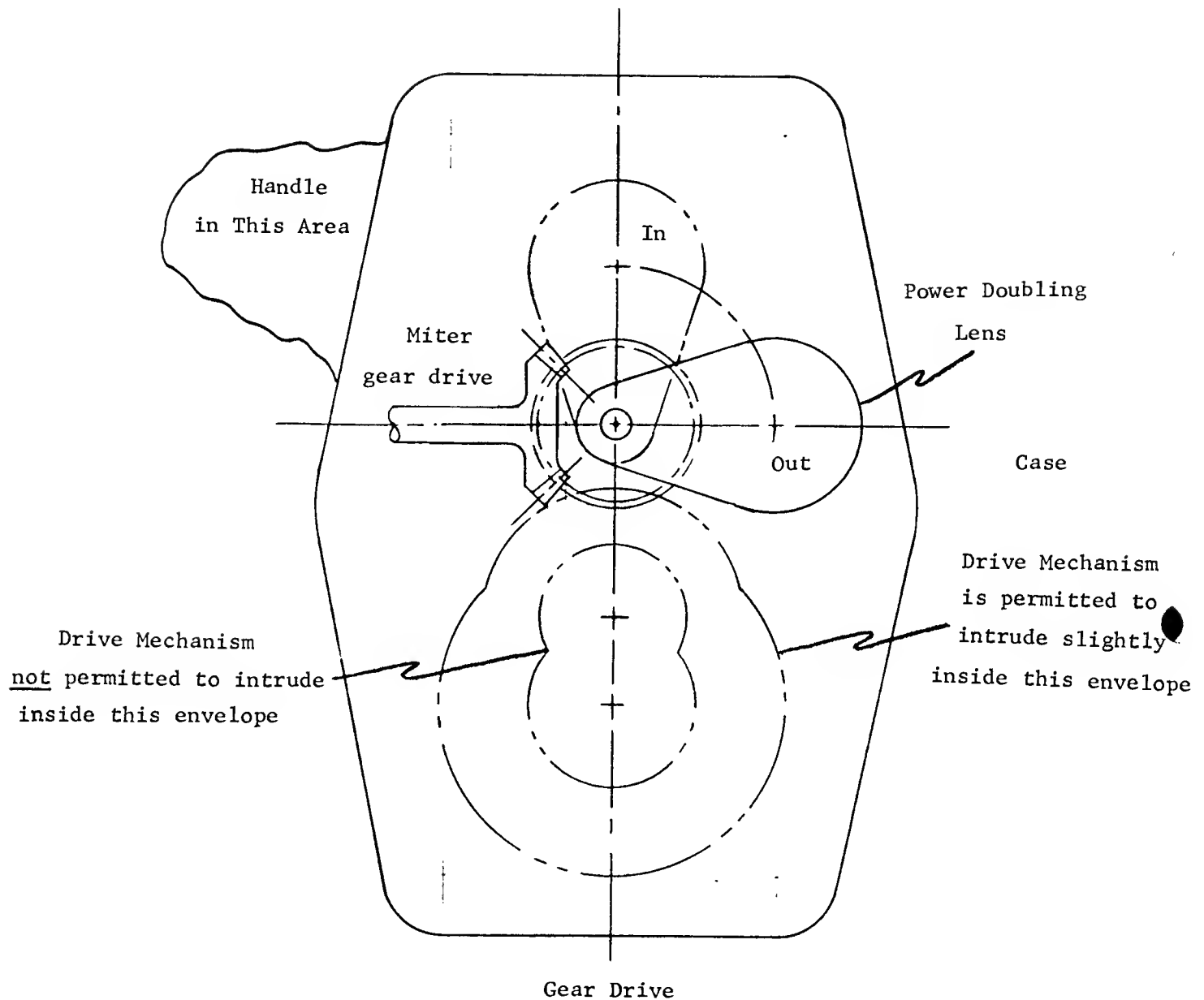
On the basis of these calculations (and the curves plotted from them), the published curves in the available literature, and the space available in the actual product, values of γ and λ were adopted for the production model and the detailed design was started. As of this date, March 1968, the production drawings are almost ready for release. Exhibit 8a to 8d shows the drawings of the actual parts which are to be used for the current production run. Exhibit 9 is a photograph of the links.



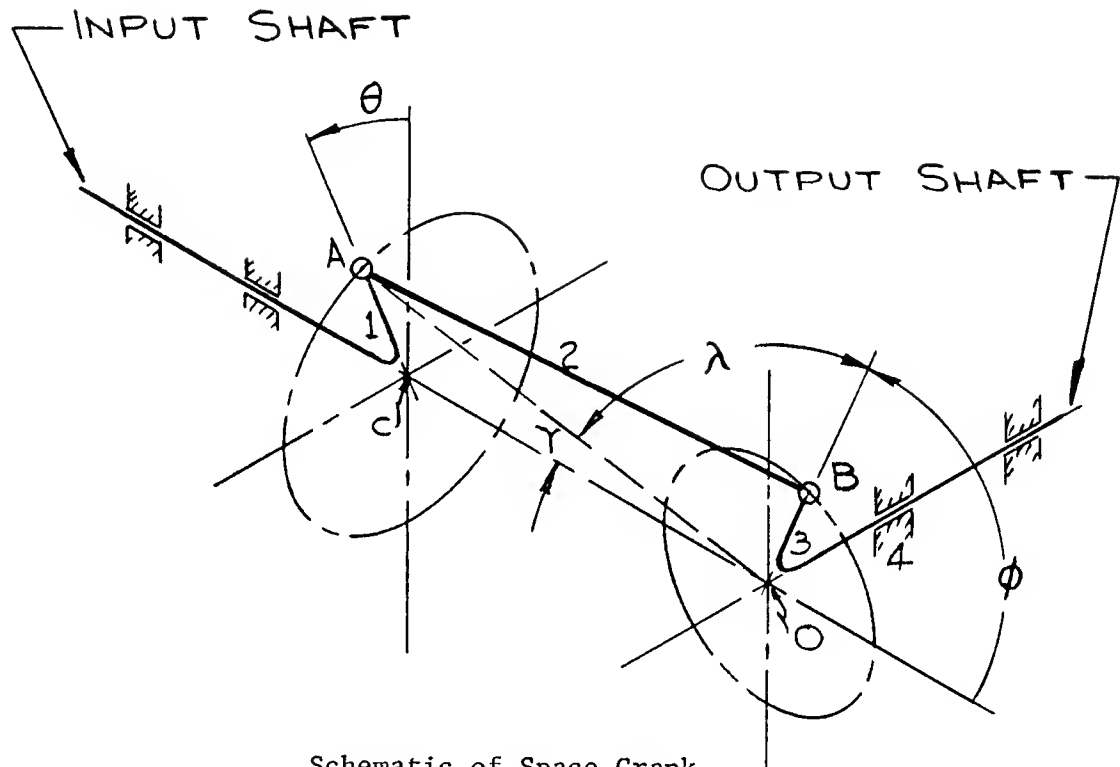
- 1 requires bi-directional movement of input
- 2 requires detent to lock the lens in "in" position
- 3 high friction and wear in slot



- 1 requires bi-directional movement of input crank
- 2 requires detent to lock the lens in "in" position
- 3 bulky & expensive
- 4 cable is susceptible to breakage



- 1 requires bi-directional movement of input
- 2 requires detent to lock the lens in "in" position
- 3 bulky & expensive



DESIGN EQUATIONS

I. DISPLACEMENT EQUATION

$$\cos \phi = \tan \gamma \cos \theta \sin \phi - \frac{\cos \lambda}{\cos \gamma}$$

γ and λ are constant for any given linkage.

II. VELOCITY EQUATION

$$\omega_{\text{output}} = \omega_{\text{input}} \left(\frac{\tan \gamma \sin \theta}{1 + \tan \gamma \cos \theta \cot \phi} \right)$$

III. ACCELERATION EQUATION (For Constant Input Angular Velocity)

$$\alpha_{\text{output}} = \frac{(\omega_i^2 + \omega_o^2 \tan \gamma \cos \theta + \omega_o \cot \phi (2 \omega_i \tan \gamma \sin \theta - \omega_o))}{1 + \tan \gamma \cos \theta \cot \phi}$$

$$\gamma = 45^\circ$$

$$a = \tan \gamma = 1.0000$$

$$\lambda = 75^\circ$$

$$b = \frac{\cos \lambda}{\cos \gamma} = \frac{.2588}{.7071} = .3660$$

I θ	II $\sin \theta$	III $\cos \theta$	IV $1+a^2 \cos^2 \theta$	V $\frac{b}{1+a^2 \cos^2 \theta}$	VI $\left[\frac{b}{1+a^2 \cos^2 \theta} \right]^2$	VII $b^2 - a^2 \cos^2 \theta$	VIII $\frac{b^2 - a^2 \cos^2 \theta}{1+a^2 \cos^2 \theta}$	IX $\left[\frac{b}{1+a^2 \cos^2 \theta} \right] - \frac{b^2 - a^2 \cos^2 \theta}{1+a^2 \cos^2 \theta}$	X $\sqrt{\text{preceding}}$
0°	0	1.0000	2.0000	.18300	.03348	-.46604	-.43302	.46650	.6830
15°	.2588	.9659	1.93296	.18934	.03584	-.79900	-.41335	.44919	.6702
30°	.5000	.8572	1.73479	.21097	.04450	-.60083	-.34634	.39084	.6251
45°	.7071	.7071	1.49999	.24400	.05954	-.36603	-.24402	.30356	.5509
60°	.8572	.5000	1.25000	.29280	.08573	-.11604	-.09283	.17856	.4225
75°	.9659	.2588	1.06697	.34302	.11766	.06698	.06277	.05489	.2347
90°	1.0000	0	1.00000	.36600	.13396	.133956	.133956	.00000	.0000
105°	.9659	-.2588	1.06697	.34302	.11766	.06698	.06277	.05489	.4225
120°	.8572	-.5000	1.25000	.29280	.08573	-.11604	-.09283	.17856	.4225
135°	.7071	-.7071	1.49999	.24400	.05954	-.36603	-.24402	.30356	.5509
150°	.5000	-.8572	1.73479	.21097	.04450	-.60083	-.34634	.39084	.6251
165°	.2588	-.9659	1.93296	.18934	.03584	-.79900	-.41335	.44919	.6702
180°	0	-1.0000	2.00000	.18300	.03348	-.46604	-.43302	.46650	.6830

$\cos \phi =$ $\cos \lambda = \cos \Sigma$	ϕ	$\Delta \phi$
.50000	60°	0
.48087	61°-16'	1°-16'
.41420	65°-32'	5°-32'
.30696	72°-08'	12°-08'
.12976	82°-33'	22°-33'
-.10874	96°-14'	36°-14'
-.36600	111°-28'	51°-28'
-.57730	125°-16'	65°-16'
-.71536	135°-40'	75°-40'
-.79496	142°-38'	82°-38'
-.83614	146°-44'	86°-44'
-.85955	149°-15'	89°-15'
-.86600	150°	90°

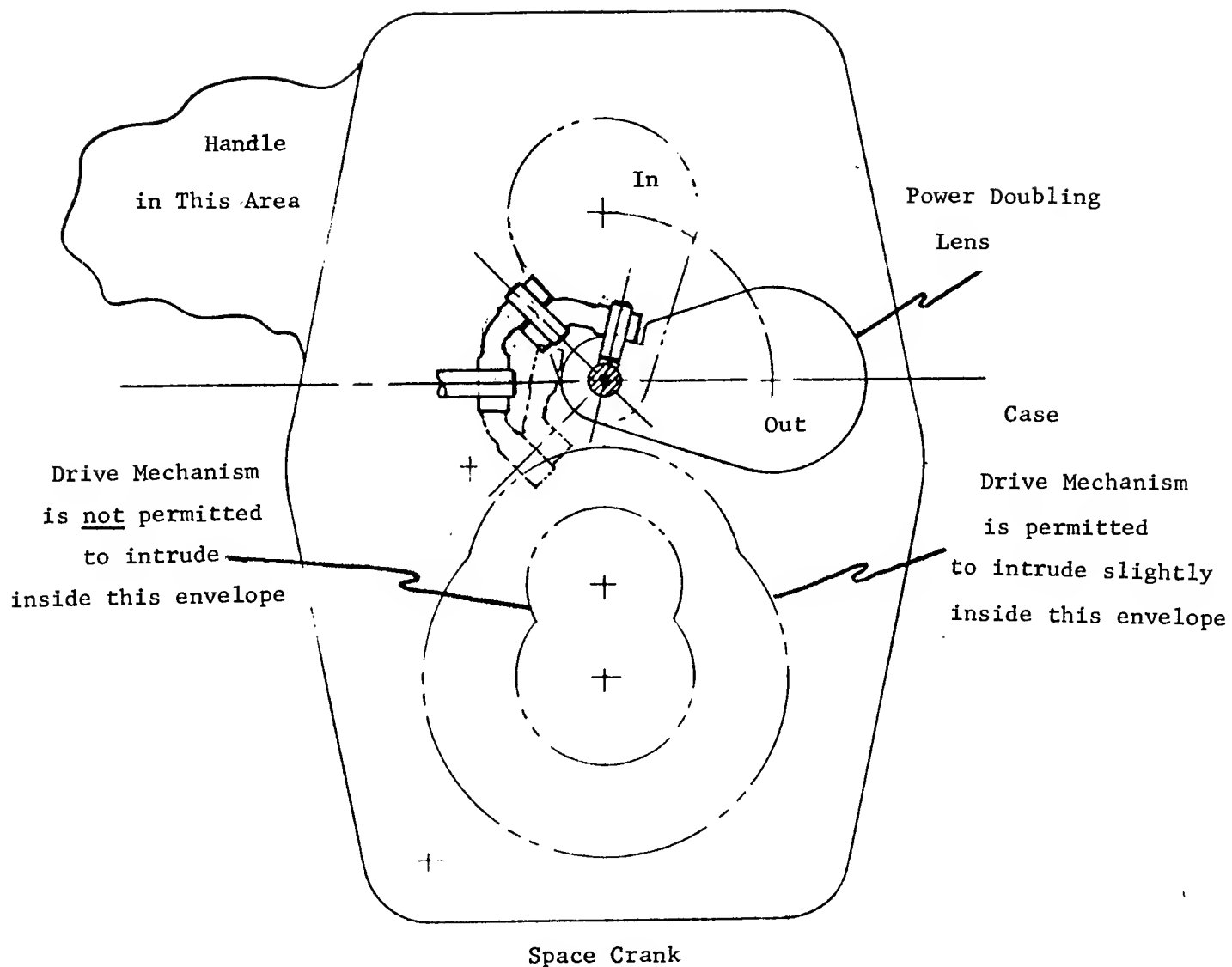
$$\gamma = 45^\circ$$

$$a = \tan \gamma = 1.0000$$

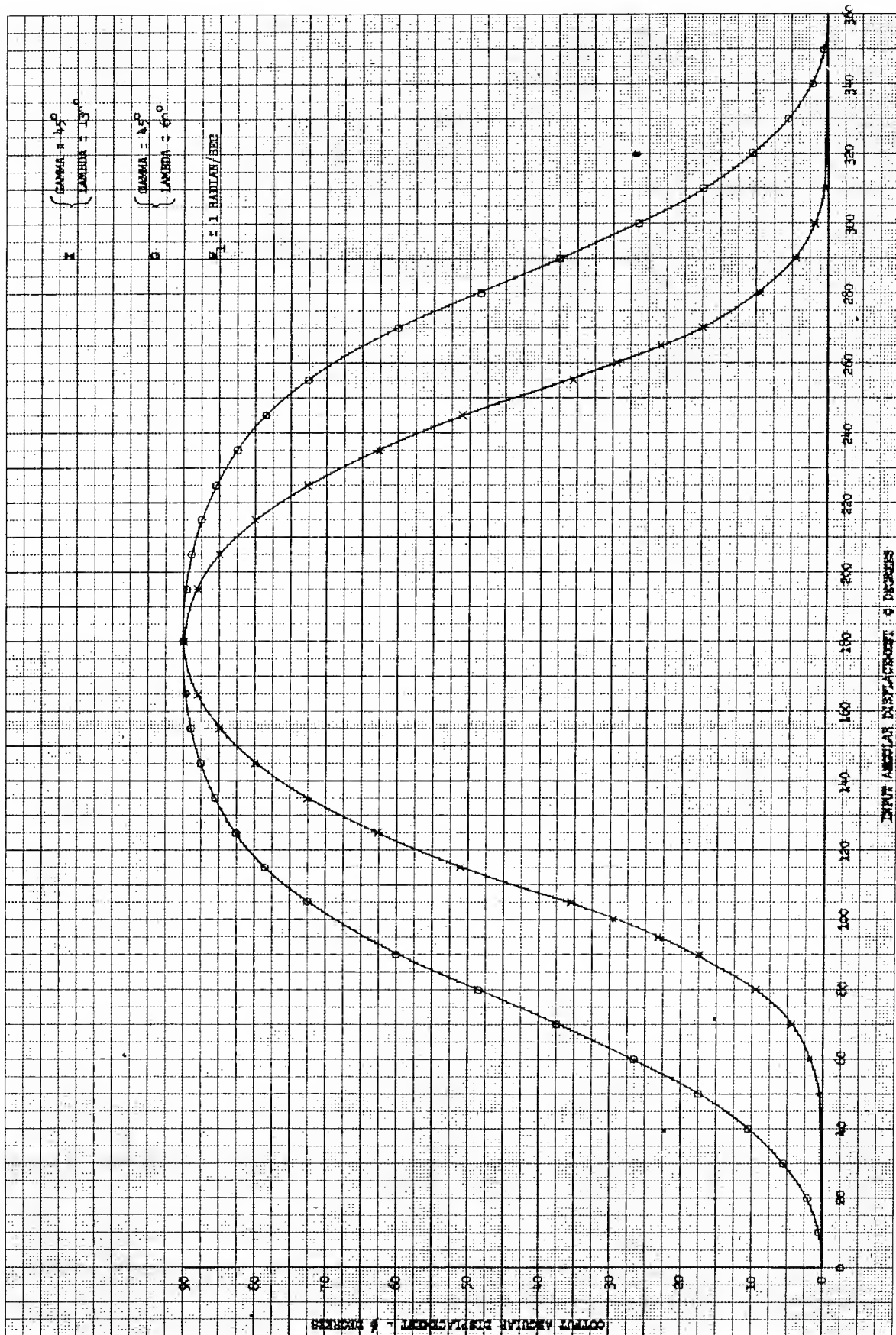
$$\lambda = 105^\circ$$

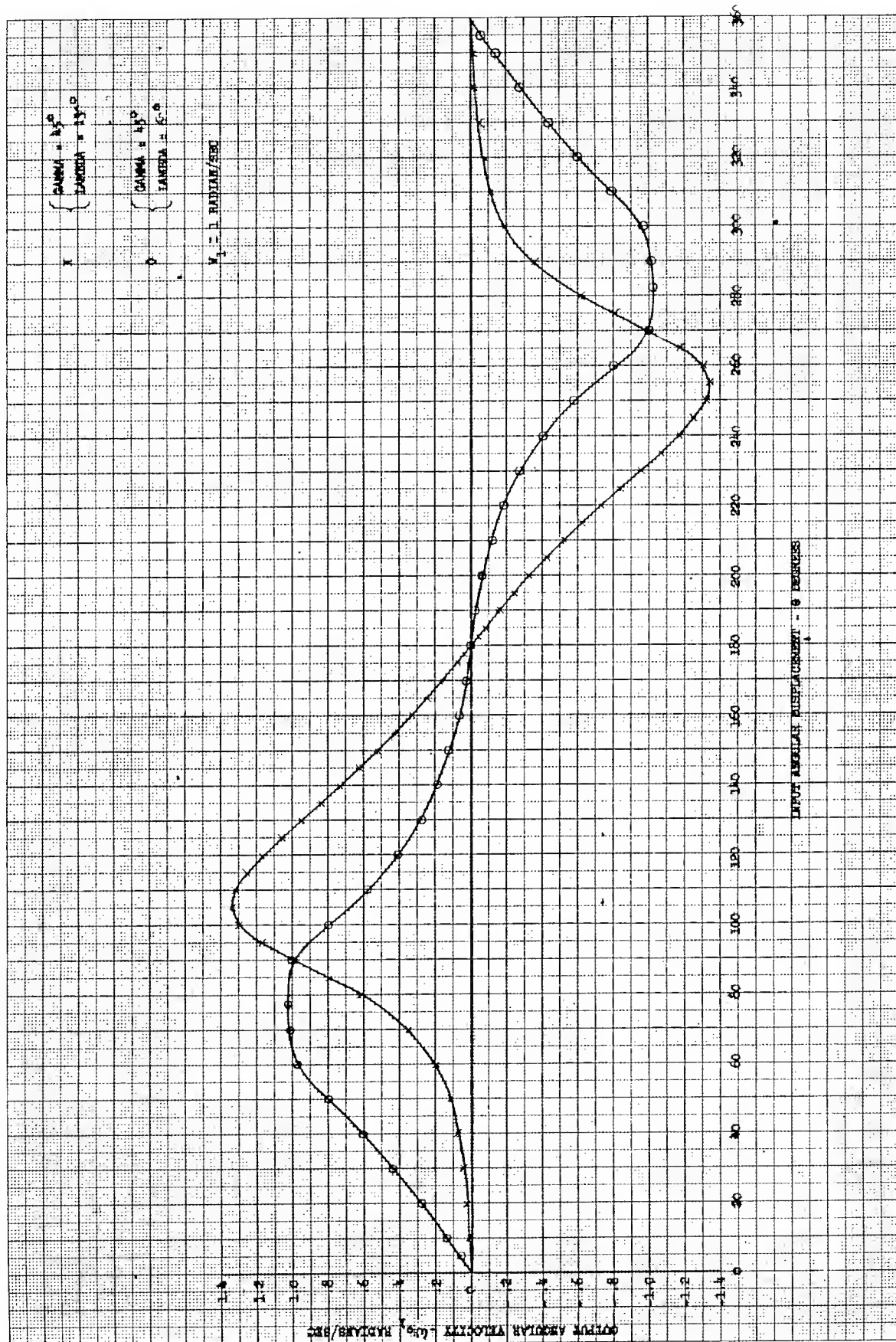
$$b = \frac{\cos \lambda}{\cos \gamma} = \frac{-0.2598}{.7071} = -.36600$$

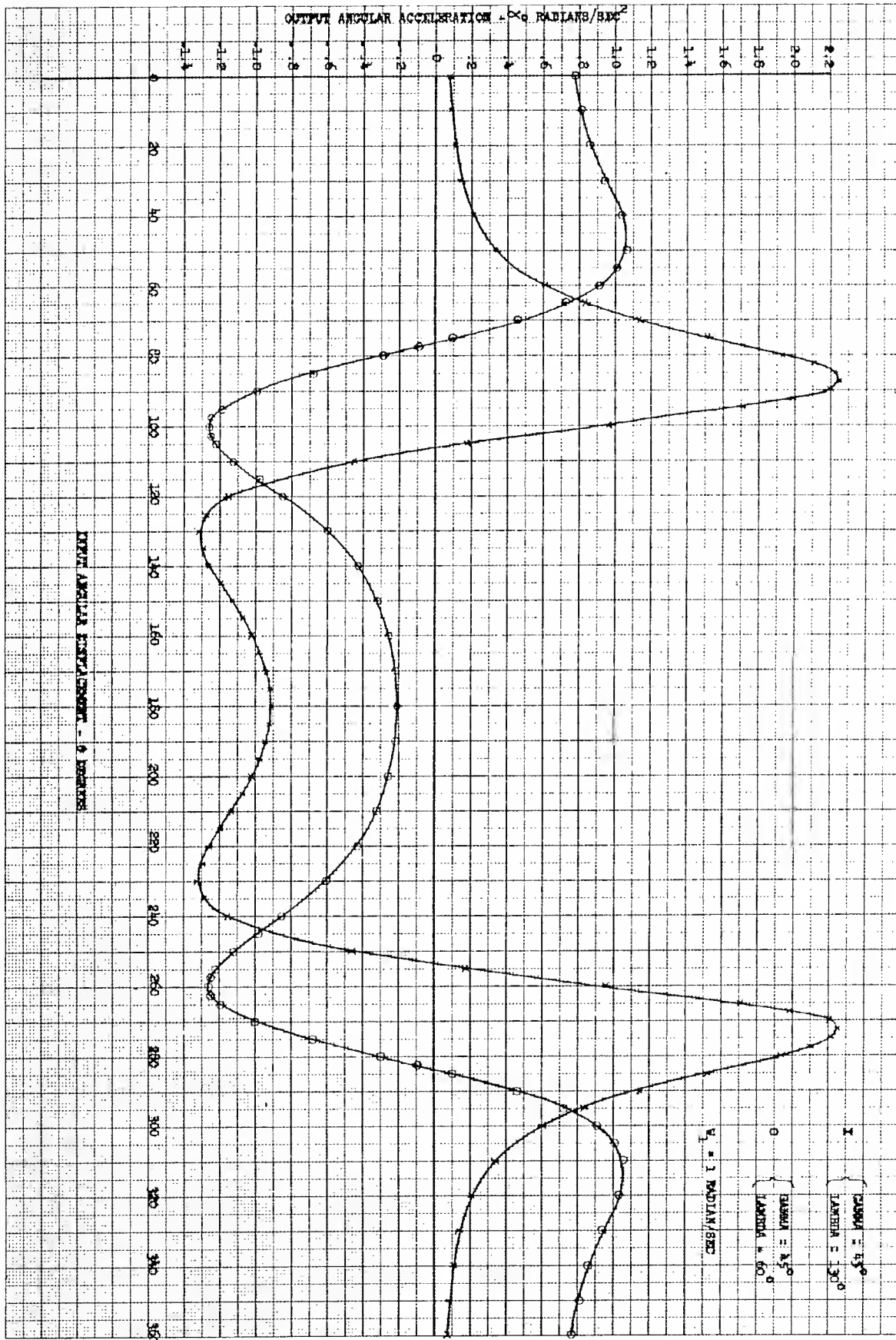
<u>XIV</u>	<u>XV</u>	<u>XVI</u>	<u>XVII</u>	<u>XVIII</u>	<u>XIX</u>	<u>XX</u>	<u>XXI</u>
$\frac{b}{1 + a \cos \theta}$						$\cos \phi =$ $-\cos \lambda = -\cos \Sigma$	ϕ
.18300	IDENTICAL TO COLUMN I					.8660	30°
-.18934						.85955	30°-45'
-.21097		IDENTICAL TO COLUMN I				.83614	33°-16'
-.24400		<u>VII</u>	IDENTICAL TO COLUMN I			.79496	37°-22'
-.29280		<u>XIV</u>	IDENTICAL TO COLUMN I			.71536	44°-20'
-.34302				<u>II</u>	IDENTICAL TO COLUMN I	.57730	54°-44'
-.36600						.36600	68°-32'
-.34302						.10874	83°-46'
-.29280						-.12976	97°-33'
-.24400						-.30696	107°-13'
-.21097						-.41420	114°-28'
-.18934						-.48087	118°-44'
-.18300						-.50000	120°



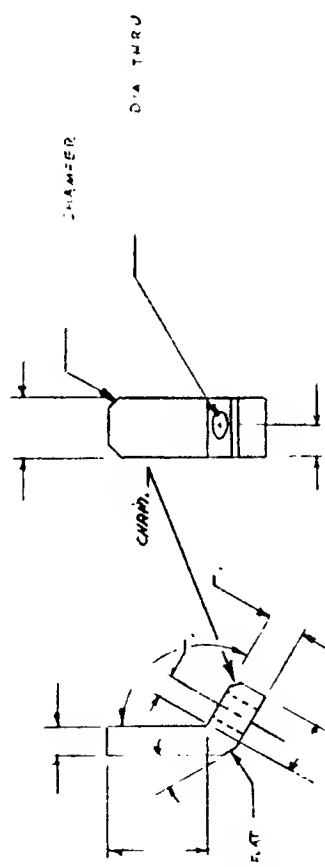
- 1 requires uni-directional movement on input crank
- 2 linkage "locks" lens in "in" position and does not require an additional detent
- 3 pin joints are rugged
- 4 inexpensive








REVISIONS			
SYM	DESCRIPTION	EO	DATE APPROVED
1			



ITEM NO.	QTY REQD	PART NUMBER	DESCRIPTION	SPECIFICATION
LIST OF MATERIALS				
			 MARK SYSTEMS, INC. 2791 SAN YSIDRO WAY SANTA CLARA, CALIFORNIA	
			POWER DOUBLER LINK	

DRAWN	DATE	APPROVED PROJECT	DATE	APPROVED	DATE	MATERIAL	FINISH

APPLICATION	QTY REQ'D	GENERAL MANUFACTURING INSTRUCTIONS S.P. 1.4.0 ARE PART OF THIS DRAWING

UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ON	FRACTIONS	DECIMALS	ANGLES	CODE IDENT NO. SIZE	SCALE	SHEET / OF /
				B		

Power Doubler Link

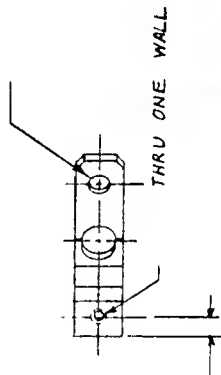
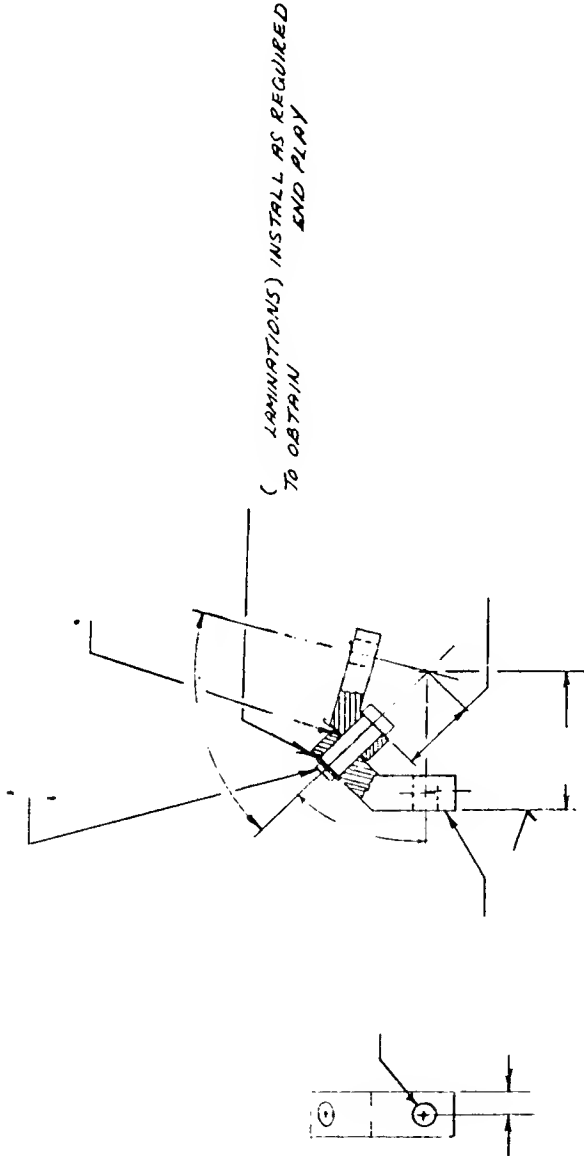
REVISIONS				DESCRIPTION		E.O.		DATE		APPROVED	
SYM	NO.	DATE	BY	DESCRIPTION	E.O.	DATE	APPROVED				

TECHNICAL DRAWING

ITEM NO.	QTY REQD	PART NUMBER	DESCRIPTION	SPECIFICATION
LIST OF MATERIALS				
			MARK SYSTEMS, INC.	
			3000 SAN VIDUO WAY	
			SANTA CLARA, CALIFORNIA	
			POWER DOUBLER LINK	
			CODE IDENT NO	SIZE
				B
			SCALE	
			SHEET / OF /	

Power Doubler Link

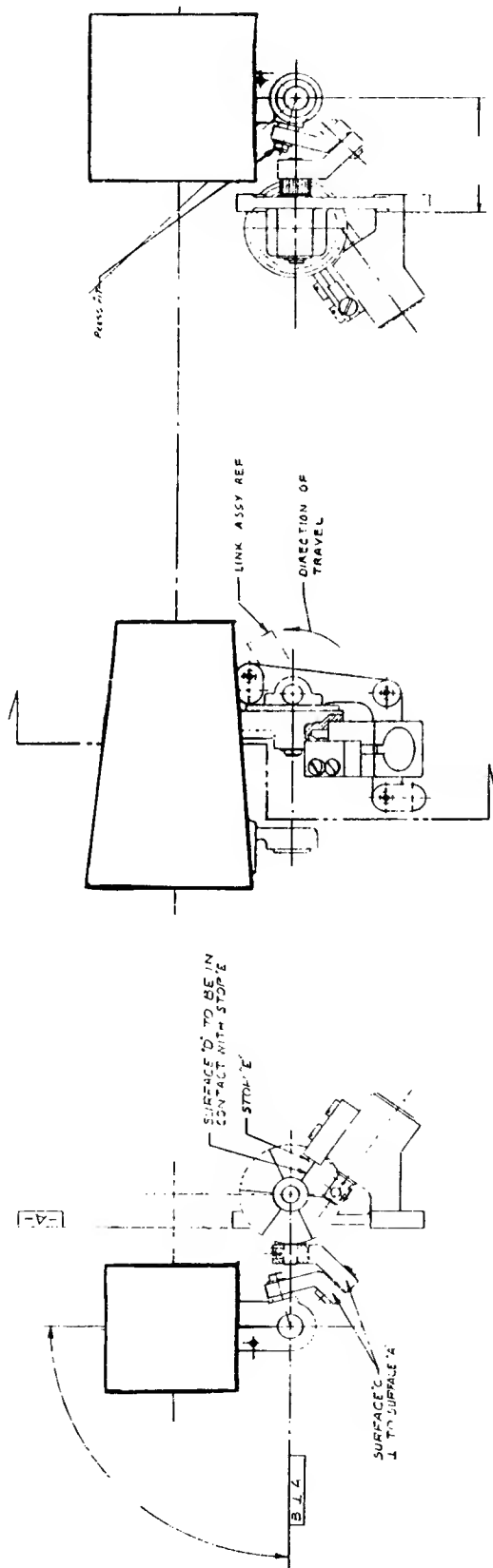
REVISIONS			
SYM	DESCRIPTION	E.C.O.	DATE APPROVED
1			



PRODUCTION CONTROL

ITEM NO.	QTY REQD	PART NUMBER	DESCRIPTION	SPECIFICATION
LIST OF MATERIALS				
DRAWN	DATE	MARK SYSTEMS, INC. 2010 SAN YSIDRO WAY SANTA CLARA, CALIFORNIA		
CHECKED	DATE			
APPROVED PROJECT	DATE	LINK MACHINING AND ASSY		
APPROVED	DATE			
MATERIAL		CODE IDENT NO	SIZE	
FINISH			C	
NOTED		SCALE	SHEET 1 OF 1	

NOTES 1.



REFERENCE

[illegible]



Photograph of Power Doubler Link Assembly

This case history may be used in both undergraduate as well as graduate courses in kinematics.

The design problem described here affords an opportunity to discuss the following:

1. In solving his problem, the author implicitly sought a mechanism which would produce an oscillatory motion from a uni-directional input motion. Plane four-bar-linkages (i.e., crank-rocker mechanisms) can be designed to satisfy this requirement. Grashof's rule states that if the sum of the shortest and longest link is less than that of the other two, and the shortest link is the crank, then a crank-rocker mechanism results. The students might be assigned the problem of finding a plane four-bar-linkage with the desired maximum output angle. Their solution might then be evaluated in terms of the comparative velocity and acceleration characteristics as well as the feasibility of building a linkage in the space available for it.
2. Appendix A shows how the author rearranged the equations shown on Exhibit 1 to facilitate computation. This could be used to illustrate the technique of solving an implicit trigonometric equation in which a trigonometric function of the dependent variable appears on both sides of the equation in non factorable form.
3. For those groups which have had some experience in writing computational programs this case history provides an opportunity to write such a program. Appendix B shows a program written in Fortran IV for IBM System 360. This program is based on the equations in Appendix A and it computes values of displacement, velocity, and acceleration for a wide range of linkage parameters. A typical print-out is included in Appendix B.

4. The curves of displacement, velocity, and acceleration present an opportunity to use the techniques of graphical differentiation or integration. A worthy exercise, for example would call for verification of the velocity and acceleration curves - given the displacement curve.

5. A photograph of the production fixture used by Mark Systems is included in Appendix C to demonstrate the simple, low cost technique which was used to manufacture the actual links. While no actual data are available, the students might be asked to evaluate this approach for a production lot of one thousand. They may also be asked to present alternative methods for the same quantity and/or quantities of ten thousand and one hundred thousand.

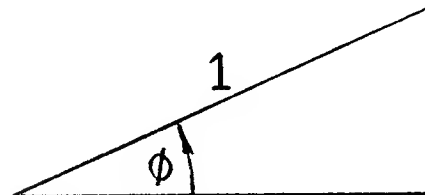
References

- R. S. Hartenberg and J. Denavit, Kinematic Synthesis of Linkages, McGraw Hill, 1964.
- R. T. Hinkle, Kinematics of Machines, Prentice Hall, 1960.
- K. H. Hunt, Mechanisms and Motion, John Wiley, 1959.
- F. R. E. Crossley, "3-D Mechanisms", Machine Design, 1955.
- L. Harrisberger, "Technique for Synthesizing Space Crank Mechanisms", Machine Design, 1964.
- J. O. Predale and A. B. Hulse Jr., "Introducing the Space Crank", Product Engineering, 1959.

DISPLACEMENT EQUATION

$$\cos \phi = \tan \gamma \cos \theta \sin \phi - \frac{\cos \lambda}{\cos \gamma}$$

$$\cos \phi = a \cos \theta \sin \phi - b$$



$$[1 - \cos^2 \phi]^{1/2} = \sin \phi$$

$$\cos \phi = a \cos \theta \sin \phi - b$$

$$\cos \phi = a \cos \theta [1 - \cos^2 \phi]^{1/2} - b$$

$$(\cos \phi + b)^2 = a^2 \cos^2 \theta [1 - \cos^2 \phi]$$

$$\cos^2 \phi + 2b \cos \phi + b^2 = a^2 \cos^2 \theta - a^2 \cos^2 \theta \cos^2 \phi$$

$$\cos^2 \phi (1 + a^2 \cos^2 \theta) + 2b \cos \phi + [b^2 - a^2 \cos^2 \theta] = 0$$

$$\cos^2 \phi + \left[\frac{2b}{1 + a^2 \cos^2 \theta} \right] \cos \phi + \left[\frac{b^2 - a^2 \cos^2 \theta}{1 + a^2 \cos^2 \theta} \right] = 0$$

$$\cos \phi = \frac{- \left[\frac{2b}{1 + a^2 \cos^2 \theta} \right] \pm \sqrt{\left[\frac{2b}{1 + a^2 \cos^2 \theta} \right]^2 - 4 \left[\frac{b^2 - a^2 \cos^2 \theta}{1 + a^2 \cos^2 \theta} \right]}}{2}$$

$$\cos \phi = - \left[\frac{b}{1 + a^2 \cos^2 \theta} \right] \pm \sqrt{\left[\frac{b}{1 + a^2 \cos^2 \theta} \right]^2 - \left[\frac{b^2 - a^2 \cos^2 \theta}{1 + a^2 \cos^2 \theta} \right]}$$

VELOCITY EQUATION

$$\frac{\omega_o}{\omega_i} = \left(\frac{a \sin \theta}{1 + a \cos \theta \cot \phi} \right)$$

$$\text{for } \omega_i = 1 \text{ rad/sec} : \omega_{o1} = \left(\frac{a \sin \theta}{1 + a \cos \theta \cot \phi} \right)$$

$$\text{or } \omega_o = \omega_i \times \omega_{o1}$$

ACCELERATION EQUATION

$$\alpha_o = \frac{(\omega_i^2 + \omega_o^2) a \cos \theta + \omega_o \cot \phi (2 a \omega_i \sin \theta - \omega_o)}{1 + a \cos \theta \cot \phi}$$

$$\alpha_o = \frac{(\omega_i^2 + \omega_i^2 \omega_{o1}^2) a \cos \theta + \omega_i \omega_{o1} \cot \phi (2 a \omega_i \sin \theta - \omega_i \omega_{o1})}{1 + a \cos \theta \cot \phi}$$

$$= \frac{\omega_i^2 (1 + \omega_{o1}^2) a \cos \theta + \omega_i^2 \omega_{o1} \cot \phi (2 a \sin \theta - \omega_{o1})}{1 + a \cos \theta \cot \phi}$$

$$= \omega_i^2 \left[\frac{(1 + \omega_{o1}^2) a \cos \theta + \omega_{o1} \cot \phi (2 a \sin \theta - \omega_{o1})}{1 + a \cos \theta \cot \phi} \right]$$

LEVEL 13 (23 MAY 67)

```
COMPILER OPTIONS - NAME= MAIN,OPT=00,LINECNT=50,SOURCE,EBCOIC,NDLIST,NODECK,LOAD,MAP,NOEDIT,ID
```

```
C
C      PROGRAM CRANK
C
    DIMENSION MAMMA(15), GAMMR(15), LAMDA(35), AMOR(35), AAMOA(35)
    100 FORMAT (I11,'/',X,X,GAMMA=' ,I3,' DEG LAMDA=' ,I4,' DEG TAN(GAMMA
        1)=',F6.4,'=A COS(LAMDA)/COS(GAMMA)=' ,F6.4,'=B OMEGA IN=1 RAD/S
        2EC ALPHA IN=0')
    101 FORMAT (/ ,5X,'THETA,DEG THETA,RAO PHI,DEG PHI,RAO DME
        1GA(OUT) ALPHA(OUT)',/,5X,6I'-----',2X))
    102 FORMAT (5X,F6.1,F13.4,F14.4,F12.4,F12.4,F12.4)
        1ATOR=.0174533
        2DO 1 I=1,11
            3MAMMA(I)=10+5*(I)
            4GAMMR(I)=ATOR*MAMMA(I)
            5A=TAN(GAMMR(I))
            6JMIN = 5+I
            7JMAX = 33-I
            8DO 2 J=JMIN,JMAX
                9SN = 1.
                10LAMDA(J) = 5*J
                11AAMOA(J) = LAMDA(J)
                12ANDR(J)=ATOR*AAMOA(J)
                13B=COS(AMOR(J))/COS(GAMMR(I))
                14K=0
                15LMAX = 36
                16THETA = 0.
                17WRITE (6,100) MAMMA(I),LAMDA(J),A,B
                18WRITE (6,101)
                19DO 4 L=L,LMAX
                    20THETR=ATOR*THETA
                    21D=1.+(A*COS(THETR))*2
                    22WWW=-B/D+SQR((-B/D)**2-(A*COS(THETR))**2)/O)*SN
                    23PHI = ARCS(WWW)
                    24PHIA=PHI/ATOR
                    25OW=1.+A*COS(THETR)*CUTAN(PHI)
                    26WUWI=A*SIN(THETR)/OW
                    27ALPHA=(11.+WUWI**2)*A*COS(THETR)+WUWI*COTAN(PHI))*(2.*A*SIN(THETR)-
                        281WUWI)/DW
                    29WRITE (6,102) THETA,THETR,PHIA,PHI,WUWI,ALPHA
                    30THETA = THETA + 2.5
                    31K=K+1
                    32IF(K.GT.1) GO TO 2
                    33LMAX = 37
                    34SN = -1.
                    35GO TO 3
                36CONTINUE
                371 CONTINUE
                38STOP
                39END
C
C      ISN 0002
C      ISN 0003
C
C      ISN 0004
C      ISN 0005
C      ISN 0006
C      ISN 0007
C      ISN 0008
C      ISN 0009
C      ISN 0010
C      ISN 0011
C      ISN 0012
C      ISN 0013
C      ISN 0014
C      ISN 0015
C      ISN 0016
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C      ISN 0023
C      ISN 0024
C      ISN 0025
C      ISN 0026
C      ISN 0027
C      ISN 0028
C      ISN 0029
C      ISN 0030
C      ISN 0031
C      ISN 0032
C
C      ISN 0033
C      ISN 0034
C      ISN 0035
C      ISN 0036
C      ISN 0037
C      ISN 0038
C      ISN 0039
C      ISN 0040
C      ISN 0041
C      ISN 0042
C      ISN 0043
C      ISN 0044
```

GAMMA= 15 DEG				LAMBDA=				TAN(GAMMA)=0.2679=A				OMEGA IN=1 RAD/SEC				ALPHA IN=0			
THETA, DEG	THETA, RAD	PHI, DEG	PHI, RAD	COS(LAMBDA)/COS(GAMMA)=0.8966=B	OMEGA (OUT)	ALPHA (OUT)	ALPHA (IN)	THETA, DEG	THETA, RAD	PHI, DEG	PHI, RAD	COS(LAMBDA)/COS(GAMMA)=0.8966=B	OMEGA (OUT)	ALPHA (OUT)	ALPHA (IN)	THETA, DEG	THETA, RAD	PHI, DEG	PHI, RAD
0.0	0.0	134.9959	2.3502	0.0	0.0	0.3660	0.3660	0.0	0.0	134.9959	2.3502	0.0	0.0	0.3660	0.3660	0.0	0.0	134.9959	2.3502
2.5	0.0436	135.0199	2.3505	0.0160	0.0160	0.3656	0.3656	2.5	0.0436	135.0199	2.3505	0.0160	0.0160	0.3656	0.3656	2.5	0.0436	135.0199	2.3505
5.0	0.0873	135.0790	2.3576	0.0319	0.0319	0.3642	0.3642	5.0	0.0873	135.0790	2.3576	0.0319	0.0319	0.3642	0.3642	5.0	0.0873	135.0790	2.3576
7.5	0.1309	135.1792	2.3593	0.0477	0.0477	0.3620	0.3620	7.5	0.1309	135.1792	2.3593	0.0477	0.0477	0.3620	0.3620	7.5	0.1309	135.1792	2.3593
10.0	0.1745	135.3182	2.3617	0.0635	0.0635	0.3588	0.3588	10.0	0.1745	135.3182	2.3617	0.0635	0.0635	0.3588	0.3588	10.0	0.1745	135.3182	2.3617
12.5	0.2182	135.4964	2.3649	0.0790	0.0790	0.3547	0.3547	12.5	0.2182	135.4964	2.3649	0.0790	0.0790	0.3547	0.3547	12.5	0.2182	135.4964	2.3649
15.0	0.2618	135.7132	2.3686	0.0944	0.0944	0.3496	0.3496	15.0	0.2618	135.7132	2.3686	0.0944	0.0944	0.3496	0.3496	15.0	0.2618	135.7132	2.3686
17.5	0.3054	135.9681	2.3731	0.1095	0.1095	0.3436	0.3436	17.5	0.3054	135.9681	2.3731	0.1095	0.1095	0.3436	0.3436	17.5	0.3054	135.9681	2.3731
20.0	0.3491	136.2600	2.3762	0.1244	0.1244	0.3366	0.3366	20.0	0.3491	136.2600	2.3762	0.1244	0.1244	0.3366	0.3366	20.0	0.3491	136.2600	2.3762
22.5	0.3927	136.5950	2.3839	0.1389	0.1389	0.3286	0.3286	22.5	0.3927	136.5950	2.3839	0.1389	0.1389	0.3286	0.3286	22.5	0.3927	136.5950	2.3839
25.0	0.4363	136.9548	2.3903	0.1530	0.1530	0.3196	0.3196	25.0	0.4363	136.9548	2.3903	0.1530	0.1530	0.3196	0.3196	25.0	0.4363	136.9548	2.3903
27.5	0.4800	137.3544	2.3973	0.1668	0.1668	0.3096	0.3096	27.5	0.4800	137.3544	2.3973	0.1668	0.1668	0.3096	0.3096	27.5	0.4800	137.3544	2.3973
30.0	0.5236	137.7881	2.4049	0.1800	0.1800	0.2985	0.2985	30.0	0.5236	137.7881	2.4049	0.1800	0.1800	0.2985	0.2985	30.0	0.5236	137.7881	2.4049
32.5	0.5672	138.2542	2.4130	0.1923	0.1923	0.2863	0.2863	32.5	0.5672	138.2542	2.4130	0.1923	0.1923	0.2863	0.2863	32.5	0.5672	138.2542	2.4130
35.0	0.6109	138.7515	2.4217	0.2050	0.2050	0.2732	0.2732	35.0	0.6109	138.7515	2.4217	0.2050	0.2050	0.2732	0.2732	35.0	0.6109	138.7515	2.4217
37.5	0.6545	139.2780	2.4309	0.2166	0.2166	0.2589	0.2589	37.5	0.6545	139.2780	2.4309	0.2166	0.2166	0.2589	0.2589	37.5	0.6545	139.2780	2.4309
40.0	0.6981	139.8340	2.4406	0.2276	0.2276	0.2436	0.2436	40.0	0.6981	139.8340	2.4406	0.2276	0.2276	0.2436	0.2436	40.0	0.6981	139.8340	2.4406
42.5	0.7418	140.4159	2.4507	0.2379	0.2379	0.2273	0.2273	42.5	0.7418	140.4159	2.4507	0.2379	0.2379	0.2273	0.2273	42.5	0.7418	140.4159	2.4507
45.0	0.7854	141.0220	2.4613	0.2474	0.2474	0.2100	0.2100	45.0	0.7854	141.0220	2.4613	0.2474	0.2474	0.2100	0.2100	45.0	0.7854	141.0220	2.4613
47.5	0.8290	141.6524	2.4723	0.2562	0.2562	0.1918	0.1918	47.5	0.8290	141.6524	2.4723	0.2562	0.2562	0.1918	0.1918	47.5	0.8290	141.6524	2.4723
50.0	0.8727	142.3020	2.4837	0.2641	0.2641	0.1727	0.1727	50.0	0.8727	142.3020	2.4837	0.2641	0.2641	0.1727	0.1727	50.0	0.8727	142.3020	2.4837
52.5	0.9163	142.9722	2.4953	0.2712	0.2712	0.1528	0.1528	52.5	0.9163	142.9722	2.4953	0.2712	0.2712	0.1528	0.1528	52.5	0.9163	142.9722	2.4953
55.0	0.9599	143.6582	2.5073	0.2775	0.2775	0.1322	0.1322	55.0	0.9599	143.6582	2.5073	0.2775	0.2775	0.1322	0.1322	55.0	0.9599	143.6582	2.5073
57.5	1.0036	144.3585	2.5195	0.2828	0.2828	0.1110	0.1110	57.5	1.0036	144.3585	2.5195	0.2828	0.2828	0.1110	0.1110	57.5	1.0036	144.3585	2.5195
60.0	1.0472	145.0714	2.5320	0.2871	0.2871	0.0894	0.0894	60.0	1.0472	145.0714	2.5320	0.2871	0.2871	0.0894	0.0894	60.0	1.0472	145.0714	2.5320
62.5	1.0908	145.7930	2.5446	0.2906	0.2906	0.0675	0.0675	62.5	1.0908	145.7930	2.5446	0.2906	0.2906	0.0675	0.0675	62.5	1.0908	145.7930	2.5446
65.0	1.1345	146.5233	2.5573	0.2930	0.2930	0.0454	0.0454	65.0	1.1345	146.5233	2.5573	0.2930	0.2930	0.0454	0.0454	65.0	1.1345	146.5233	2.5573
67.5	1.1781	147.2581	2.5701	0.2945	0.2945	0.0233	0.0233	67.5	1.1781	147.2581	2.5701	0.2945	0.2945	0.0233	0.0233	67.5	1.1781	147.2581	2.5701
70.0	1.2217	147.9947	2.5830	0.2951	0.2951	0.0014	0.0014	70.0	1.2217	147.9947	2.5830	0.2951	0.2951	0.0014	0.0014	70.0	1.2217	147.9947	2.5830
72.5	1.2654	148.7321	2.5959	0.2946	0.2946	-0.0201	-0.0201	72.5	1.2654	148.7321	2.5959	0.2946	0.2946	-0.0201	-0.0201	72.5	1.2654	148.7321	2.5959
75.0	1.3090	149.4670	2.6087	0.2933	0.2933	-0.0411	-0.0411	75.0	1.3090	149.4670	2.6087	0.2933	0.2933	-0.0411	-0.0411	75.0	1.3090	149.4670	2.6087
77.5	1.3526	150.1972	2.6214	0.2911	0.2911	-0.0613	-0.0613	77.5	1.3526	150.1972	2.6214	0.2911	0.2911	-0.0613	-0.0613	77.5	1.3526	150.1972	2.6214
80.0	1.3963	150.9223	2.6341	0.2880	0.2880	-0.0805	-0.0805	80.0	1.3963	150.9223	2.6341	0.2880	0.2880	-0.0805	-0.0805	80.0	1.3963	150.9223	2.6341
82.5	1.4399	151.6359	2.6465	0.2841	0.2841	-0.0987	-0.0987	82.5	1.4399	151.6359	2.6465	0.2841	0.2841	-0.0987	-0.0987	82.5	1.4399	151.6359	2.6465
85.0	1.4835	152.3385	2.6588	0.2794	0.2794	-0.1156	-0.1156	85.0	1.4835	152.3385	2.6588	0.2794	0.2794	-0.1156	-0.1156	85.0	1.4835	152.3385	2.6588
87.5	1.5272	153.0330	2.6709	0.2740	0.2740	-0.1312	-0.1312	87.5	1.5272	153.0330	2.6709	0.2740	0.2740	-0.1312	-0.1312	87.5	1.5272	153.0330	2.6709

GAMMA= 15 DEG LAMDA= 30 DEG IAN(GAMMA)=0.2679=A COS(LAMDA)/COS(GAMMA)=0.8966=B OMEGA IN=1 RAD/SEC ALPHA IN=0						
IFETA, DEG	IFETA, RAD	PHI, DEG	PHI, RAD	OMEGA(OUT)	ALPHA(OUT)	
90.C	1.5708	153.7114	2.6828	0.2679	-0.1453	
92.5	1.6144	154.3741	2.6943	0.2613	-0.1579	
95.C	1.6581	155.0217	2.7056	0.2542	-0.1690	
97.5	1.7017	155.6455	2.7165	0.2466	-0.1784	
100.C	1.7453	156.2500	2.7271	0.2386	-0.1863	
102.5	1.7890	156.8380	2.7373	0.2304	-0.1926	
105.C	1.8326	157.4022	2.7472	0.2219	-0.1975	
107.5	1.8762	157.9465	2.7567	0.2132	-0.2010	
110.C	1.9194	158.4687	2.7658	0.2043	-0.2032	
112.5	1.9635	158.9670	2.7745	0.1954	-0.2043	
115.C	2.0071	159.4453	2.7828	0.1865	-0.2043	
117.5	2.0508	159.9000	2.7908	0.1776	-0.2034	
120.C	2.0944	160.3332	2.7983	0.1688	-0.2017	
122.5	2.1380	160.7450	2.8055	0.1600	-0.1993	
125.C	2.1817	161.1341	2.8123	0.1514	-0.1963	
127.5	2.2253	161.5018	2.8187	0.1429	-0.1929	
130.C	2.2689	161.8487	2.8248	0.1346	-0.1892	
132.5	2.3126	162.1745	2.8305	0.1264	-0.1853	
135.C	2.3562	162.4800	2.8358	0.1184	-0.1811	
140.C	2.4435	163.0330	2.8455	0.1030	-0.1727	
142.5	2.4871	163.2620	2.8498	0.0955	-0.1686	
145.C	2.5307	163.5114	2.8538	0.0882	-0.1645	
147.5	2.5744	163.7230	2.8575	0.0812	-0.1606	
150.C	2.6180	163.9171	2.8609	0.0742	-0.1569	
152.5	2.6616	164.0947	2.8640	0.0675	-0.1534	
155.C	2.7053	164.2543	2.8668	0.0608	-0.1502	
157.5	2.7483	164.3984	2.8693	0.0544	-0.1472	
160.C	2.7925	164.5267	2.8715	0.0480	-0.1444	
162.5	2.8362	164.6389	2.8735	0.0417	-0.1420	
165.C	2.8798	164.7351	2.8752	0.0356	-0.1399	
167.5	2.9234	164.8167	2.8766	0.0295	-0.1381	
170.C	2.9671	164.8829	2.8778	0.0235	-0.1366	
172.5	3.0107	164.9340	2.8787	0.0176	-0.1355	
175.C	3.0543	164.9710	2.8793	0.0117	-0.1346	
177.5	3.0980	164.9927	2.8797	0.0058	-0.1341	
180.C	3.1416	165.0000	2.8798	-0.0000	-0.1340	

